AIAA DPW4



Baseline Unstructured Grids

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Baseline unstructured grids

- VGrid/PostGrid 4.2
 - Volume and surface* sources used
- Cessna's contribution
 - Patching, initial sourcing and grid generation
- NASA's contribution
 - Smoothed all grids
 - Generated second set of grids with slower viscous growth rates
 - Generated cell based versions
 - Generated extra-fine grids
- Coarse, Medium, Fine and Extra-fine** grids generated
- *Surface source capability not released to public **Shahyar only able to complete NASA extra-fine grid



Grid differences

Cessna Grid

- Faster BL growth rate: R1=0.15, R2=0.06
- Volume sources at wing root
- Discrete linear sources on fuselage
- Med. grid "ifact"=1.0
- Better surface resolution

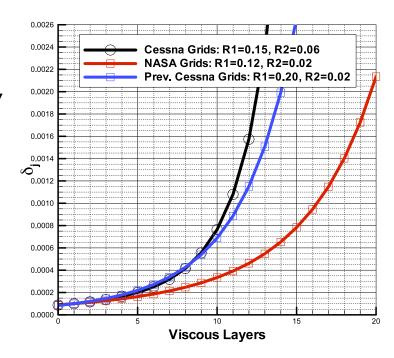
NASA Grid

- Slower BL growth rate:
 R1=0.12, R2=0.02
- Volume sources at wing/ tail TE
- Surface sources on fuselage
- Med. grid "ifact"=1.2175
- Better boundary layer resolution



Effect of growth rate parameters

- $\delta_{j} = \delta_{1} * (1 + R1 * (1 + R2)^{j-1})^{j-1}$
- Cessna grids use
 R1=0.15, R2=0.06
- NASA grids use R1=0.12, R2=0.02
- Reduced growth rate leads to increased number of points in viscous portion of grid
- "ifact" used to bring overall grid size to match that of Cessna grids





Grid statistics (node based grids)

Grid	Cessna Nodes (millions)	NASA Nodes (millions)	Cessna Faces (millions)	NASA Faces (millions)	Cessna Visc. Cells (millions)	NASA Visc. Cells (millions)
Coarse	3.55	3.66 (103%)	0.309	0.228 (74%)	15.54	18.96 (122%)
Medium Tail off	8.48	8.21 (97%)	0.554	0.418 (76%)	27.99	34.99 (125%)
Medium	9.93	10.23 (103%)	0.745	0.566 (76%)	38.09	48.60 (128%)
Fine	35.21	35.97 (102%)	2.104	1.665 (79%)	108.19	145.07 (134%)
Extra fine	NA	105.63	NA	4.045	NA	356.47

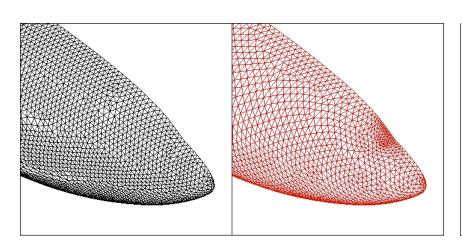


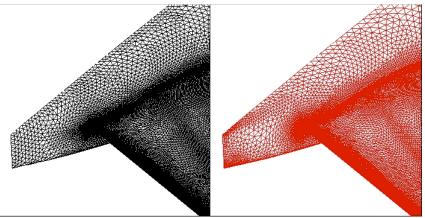
Surface resolution details

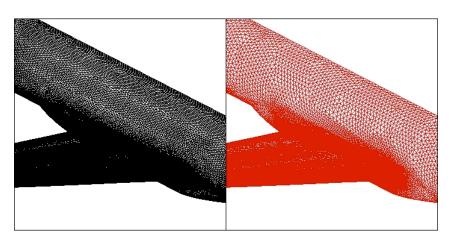
Grid	Cessna "ifact"	NASA "ifact"	Cessna Fuselage Faces	NASA Fuselage Faces	Cessna Wing/Tail Faces	NASA Wing/Tail Faces
Coarse	1.597	2.0142	39,096	17,832 (46%)	161,472/ 80,563	119,564/ 59,352 (74%)
Medium	1.0	1.2175	95,966	45,912 (48%)	399,437/ 197,983	311,198/ 152,689 (78%)
Fine	0.58	0.6845	280,971	138,885 (49%)	1,145,045/ 562,328	954,497/ 454,876 (82%)

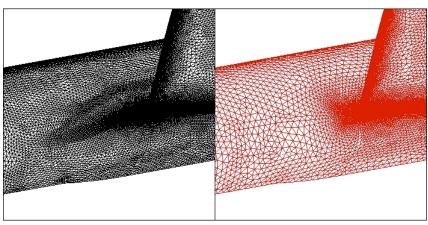


Cessna and NASA grids



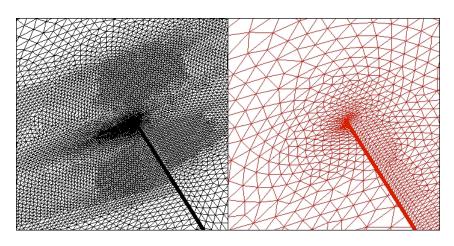


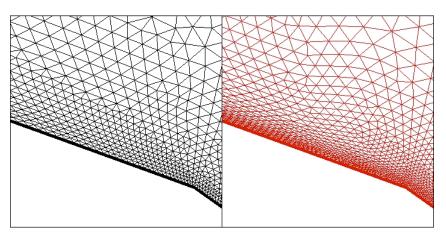


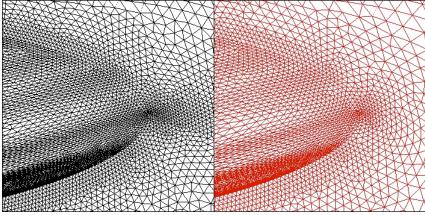




Cessna and NASA grids









Baseline unstructured grids

- NASA grids have slower viscous growth rate and more cells in the viscous layers
- Cessna grids have increased surface density, especially on the fuselage, wing, and tail
- Due to sourcing and "ifact" differences the grids represent two families of grids
- Closing fine grids could be difficult for PostGrid
- Extra-fine grid generation was very challenging for VGrid/PostGrid



Backup slides



Number of surface faces for coarse grid

Component	NASA ifact=2.0142	Cessna ifact=1.5970	% decrease due to ifact
Total	215,694	300,504	28
Outer	304	538	37
Symmetry	18,606	18,835	1
Fuselage	17,832	39,096	54
Wing	119,564	161,472	26
Tail	59,352	80,563	26



Number of surface faces for medium grid

Component	NASA ifact=1.2175	Cessna ifact = 1.0	% decrease due to ifact
Total	545710	732319	25
Outer	858	1284	33
Symmetry	35053	37649	7
Fuselage	45912	95966	52
Wing	311198	399437	22
Tail	152689	197983	23



Number of surface faces for fine grid

Component	NASA ifact=0.6845	Cessna ifact=0.58	% decrease due to ifact
Total	1629422	2083140	22
Outer	2670	3744	29
Symmetry	78494	91052	14
Fuselage	138885	280971	51
Wing	954497	1145045	17
Tail	454876	562328	19